

(張 東 斗)

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**Improved Tuned Mass Damper System for
Vibration Control of Bridges
under Moving Loads**

**Improved Tuned Mass Damper System for
Vibration Control of Bridges
under Moving Loads**

Advisor : Professor In-Won Lee

by

Dong-Doo Jang


**Department of Civil and Environmental Engineering
Korea Advanced Institute of Science and Technology**

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requirements for the degree of Master of Engineering in the
Department of Civil and Environmental Engineering**

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Approved by


Professor In-Won Lee
Major Advisor

이동하중을 받는 교량의 진동제어를 위한 개선된 동조질량감쇠시스템

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ABSTRACT

This paper presents a new vibration control of bridges under moving loads using improved tuned mass damper(TMD). Although the conventional TMD has excellent performance in the control of steady-state displacements, it has limitations in the suppression of transient displacements. This paper proposes a TMD system combined with king-post to efficiently control both steady-state and transient vibration parts of displacements.

To show the efficiency of the proposed TMD, a simple span bridge is considered. The midpoint vertical responses are investigated before and after the installation of proposed TMD. The results of the proposed TMD are also compared with those of the conventional TMD and the king-post system.

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1

1.1

가 .

가

가

가 .

1.2

가

가

가

(Passive type type)

(Active type)

(Semi-active

가

, actuator

가

가

가

mass damper)

(1998)^[1]

(Tuned (1999)^[8]

Pinkaew(1996)^[9] , Shelley(1995)^[10]

damper)

, Patten(1996)^[11]

(Active mass

(Semiactive vibration absorber)

가

1.3

King-Post

. King-Post

, King-Post

[6]

1

DB24

Moving Force Model

가

가

Den Hartog^[7]

$t + \Delta t$

(Implicit Method) Newmark β

가

(Constant-Average-Acceleration Method)

5

2

2.1

가 Moving Force 가 Bernoulli-Euler 가

(1) stringer girder 가

(2)

(3)

2.2

[1] $m, EI, l, y(x, t)$

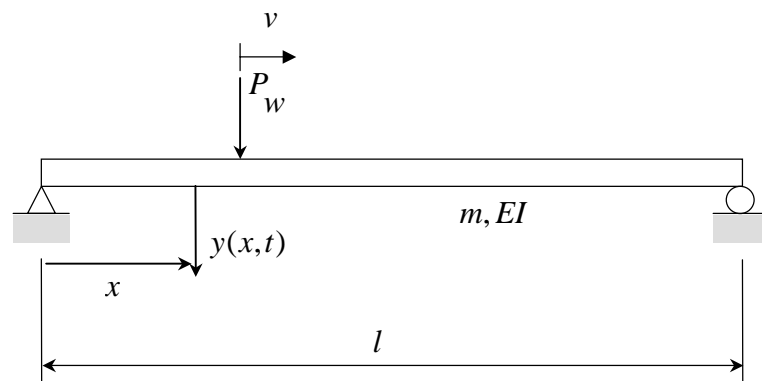
v, P_w

v

$$EI \frac{\partial^4 y}{\partial x^4} + c \frac{\partial y}{\partial t} + m \frac{\partial^2 y}{\partial t^2} = F(x, t) \quad (1)$$

$$F(x, t) = P_w \delta(x - vt) \quad (2)$$

c $F(x, t)$ 가
 Dirac delta function $\delta(x - vt)$
 가



[1]

2.3

$$EI \frac{\partial^4 y}{\partial x^4} + m \frac{\partial^2 y}{\partial t^2} = 0 \quad (3)$$

EI , m , y

$$y = \varphi(x) \sin \omega t \quad (4)$$

$$\omega \quad (4) \quad (3)$$

$$\frac{\partial^4 \varphi(x)}{\partial x^4} = \beta^4 \varphi(x) \quad (5)$$

$$\beta^4 = \frac{m \omega^2}{EI} \quad (6)$$

(eigenfunction) (5)

$$\varphi(x) = A_1 \cos \beta x + A_2 \sin \beta x + A_3 \cosh \beta x + A_4 \sinh \beta x \quad (7)$$

$$\begin{aligned} \varphi(0) = 0, & \quad \left. \frac{d^2 \varphi}{dx^2} \right|_{x=0} = 0 \\ \varphi(L) = 0, & \quad \left. \frac{d^2 \varphi}{dx^2} \right|_{x=L} = 0 \end{aligned} \quad (8)$$

A_1, A_2, A_3, A_4

$$A_1 = A_3 = A_4 = 0, \quad A_2 = A \quad (9)$$

$$\sin \beta L = 0 \quad (10)$$

A

(6), (9) (10)

$$\omega_i = \left(\frac{i\pi}{L} \right)^2 \sqrt{\frac{EI}{m}} \quad (11)$$

$$\varphi_i(x) = \sin \frac{i\pi x}{L} \quad (12)$$

2.4

2.3

(13)

$$y(x, t) = \sum_{i=1}^n q_i(t) \varphi_i(x) \quad (13)$$

(13)

(1)

$$m \sum_{i=1}^n \varphi_i(x) \ddot{q}_i(t) + c \sum_{i=1}^n \varphi_i(x) \dot{q}_i(t) + EI \sum_{i=1}^n \varphi_i^{(4)}(x) q_i(t) = F(x, t) \quad (14)$$

(14) j mode

$$\begin{aligned} \int_0^L m \varphi_j(x) \sum_{i=1}^n \varphi_i(x) \ddot{q}_i(t) dx + \int_0^L c \varphi_j(x) \sum_{i=1}^n \varphi_i(x) \dot{q}_i(t) dx \\ + \int_0^L EI \varphi_j(x) \sum_{i=1}^n \varphi_i^{(4)}(x) q_i(t) dx = \int_0^L \varphi_j(x) F(x, t) dx \end{aligned} \quad (15)$$

$$L, \varphi_i^{(4)} i 4 . \quad (15)$$

$$\ddot{q}_i(t) + 2\zeta_i \omega_i \dot{q}_i(t) + \omega_i^2 q_i(t) = \frac{1}{\int_0^L m(\varphi_j(x))^2 dx} \int_0^L \varphi_j(x) F(x, t) dx \quad (16)$$

$$\zeta_i i, \omega_i i \quad (16) \quad (2) \quad (12)$$

$$\ddot{q}_i(t) + 2\xi_i \omega_i \dot{q}_i(t) + \omega_i^2 q_i(t) = \frac{2}{mL} P_w \sin \frac{i\pi w t}{L} \quad (i = 1, 2, \dots) \quad (17)$$

$$M \ddot{q} + C \dot{q} + K q = Q \quad (18)$$

M, C, K , , ,
 Q , q .

$$M = \begin{bmatrix} 1 & 0 & \dots & 0 \\ 0 & 1 & & \vdots \\ \vdots & & \ddots & 0 \\ 0 & \dots & 0 & 1 \end{bmatrix}$$

$$C = \begin{bmatrix} 2\zeta_1 \omega_1 & 0 & \dots & 0 \\ 0 & 2\zeta_2 \omega_2 & & \vdots \\ \vdots & & \ddots & 0 \\ 0 & \dots & 0 & 2\zeta_n \omega_n \end{bmatrix}$$

$$K = \begin{bmatrix} \omega_1^2 & 0 & \dots & 0 \\ 0 & \omega_2^2 & & \vdots \\ \vdots & & \ddots & 0 \\ 0 & \dots & 0 & \omega_n^2 \end{bmatrix}$$

$$Q = \frac{2P_w}{mL} N_v , \quad q = \begin{Bmatrix} q_1 \\ q_2 \\ \vdots \\ q_n \end{Bmatrix}, \quad N_v = \begin{Bmatrix} \sin \frac{\pi vt}{L} \\ \sin \frac{2\pi vt}{L} \\ \vdots \\ \sin \frac{n\pi vt}{L} \end{Bmatrix} \quad (19)$$

$$(18) \quad t + \Delta t$$

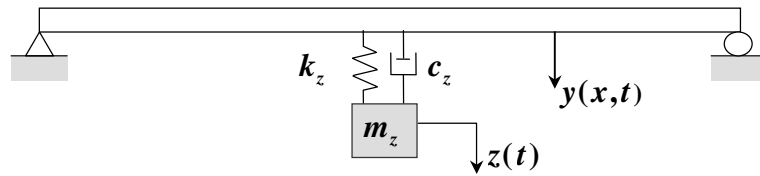
Newmark β

3

3.1

King-Post
(TMD)
King-Post 가
, King-Post 가
TMD
1 가
King-Post 가
2 가

3.2 TMD 가



[2] TMD 가

TMD 가 ,
 [2] TMD . TMD 가
 2
 TMD 가 .
 가 TMD TMD .
 (20) TMD .

$$m_z \ddot{z} + c_z (\dot{z} - \dot{y}) + k_z (z - y) = 0 \quad (20)$$

m_z, c_z, k_z TMD .
 TMD ,
 ,
 (21) .

$$F(x, t) = P_w \delta(x - vt) + (m_z g - m_z \ddot{z}) \delta(x - L/2) \quad (21)$$

(16) (21) .

$$\begin{aligned} \ddot{q}_i(t) + 2\xi_i \omega_i \dot{q}_i(t) + \omega_i^2 q_i(t) + \frac{2m_z}{mL} \sin \frac{i\pi}{2} \cdot \ddot{z}(t) \\ = \frac{2}{mL} \left[P_w \sin \frac{i\pi vt}{L} + m_z g \sin \frac{i\pi}{2} \right] \quad (i = 1, 2, \dots) \end{aligned} \quad (22)$$

(20) y (13)

$$m_z \ddot{z}(t) + c_z \dot{z}(t) + k_z z(t) - c_z \varphi_i(L/2) \dot{q}_i(t) - k_z \varphi_i(L/2) q_i(t) = 0 \quad (i = 1, 2, \dots) \quad (23)$$

$$(22) \quad (23)$$

$$(24)$$

$$M\ddot{q} + C\dot{q} + Kq = Q \quad (24)$$

$$M = \begin{bmatrix} I_{n \times n} & M_{12} \\ \mathbf{0}_{1 \times n} & m_z \end{bmatrix} \quad C = \begin{bmatrix} C_{11} & \mathbf{0}_{n \times 1} \\ C_{21} & c_z \end{bmatrix} \quad K = \begin{bmatrix} K_{11} & \mathbf{0}_{n \times 1} \\ K_{21} & k_z \end{bmatrix}$$

$$M_{12} = \frac{2m_z}{ml} N$$

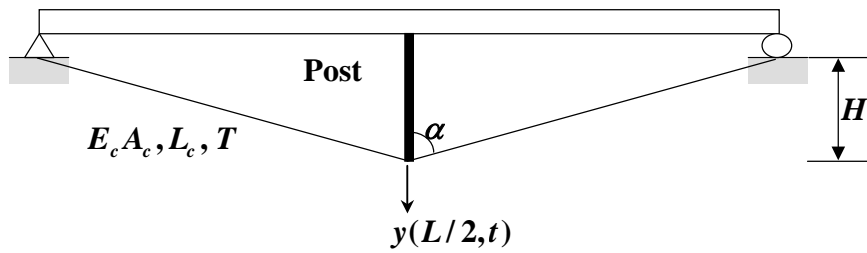
$$C_{11} = \text{diag}(2\zeta_i \omega_i) \quad C_{21} = -c_z N^T \quad (i = 1, 2, \dots, n)$$

$$K_{11} = \text{diag}(\omega_i^2) \quad K_{21} = -k_z N^T \quad (i = 1, 2, \dots, n)$$

$$Q = \frac{2}{ml} (P_w N_v + m_z g N)$$

$$q = \begin{Bmatrix} q_1 \\ q_2 \\ \vdots \\ q_n \\ z \end{Bmatrix} \quad N = \begin{Bmatrix} \sin \frac{\pi}{2} \\ \sin \frac{2\pi}{2} \\ \vdots \\ \sin \frac{n\pi}{2} \end{Bmatrix} \quad N_v = \begin{Bmatrix} \sin \frac{\pi vt}{L} \\ \sin \frac{2\pi vt}{L} \\ \vdots \\ \sin \frac{n\pi vt}{L} \end{Bmatrix} \quad (25)$$

3.3 King-Post 가



[3] King-Post 가

King-Post TMD 가 가 King-Post

. King-Post Post
 . Post 가
 $y(L/2, t)$. King-Post 가

2 King-Post

$F(x, t)$

$$F(x, t) = P_w \delta(x - vt) - P_k \delta(x - L/2) \quad (26)$$

P_k King-Post 가

T , Post α ,

Δ

$$\Delta = y(L/2, t) \cdot \cos \alpha = \frac{TL_c}{E_c A_c} = \frac{TH}{E_c A_c \cos \alpha} \quad (27)$$

$$P_k = 2T \cos \alpha \quad (28)$$

$$(27) \quad T \quad (28)$$

$$P_k = \frac{2E_c A_c}{H} \cos^3 \alpha \cdot y(L/2, t) = \frac{2E_c A_c}{H} \cos^3 \alpha \cdot \sin \frac{i\pi}{2} q_i(t) \quad (i = 1, 2, \dots, n) \quad (29)$$

$$(29) \quad (26) \quad (16)$$

$$\begin{aligned} \ddot{q}_i(t) + 2\xi_i \omega_i \dot{q}_i(t) + \left(\omega_i^2 + \frac{4E_c A_c}{mLH} \cos^3 \alpha \cdot \sin \frac{i\pi}{2} \sin \frac{j\pi}{2} \right) q_i(t) \\ = \frac{2}{mL} P_w \sin \frac{i\pi vt}{L} \quad (i, j = 1, 2, \dots) \end{aligned} \quad (30)$$

$$(31)$$

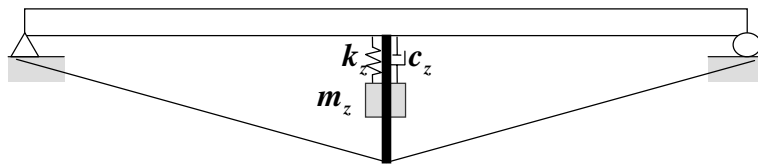
$$M\ddot{q} + C\dot{q} + Kq = Q \quad (31)$$

$$M = \begin{bmatrix} 1 & 0 & \cdots & 0 \\ 0 & 1 & & \vdots \\ \vdots & & \ddots & 0 \\ 0 & \cdots & 0 & 1 \end{bmatrix} \quad C = \begin{bmatrix} 2\zeta_1\omega_1 & 0 & \cdots & 0 \\ 0 & 2\zeta_2\omega_2 & & \vdots \\ \vdots & & \ddots & 0 \\ 0 & \cdots & 0 & 2\zeta_n\omega_n \end{bmatrix}$$

$$K = \begin{bmatrix} \omega_1^2 & 0 & \cdots & 0 \\ 0 & \omega_2^2 & & \vdots \\ \vdots & & \ddots & 0 \\ 0 & \cdots & 0 & \omega_n^2 \end{bmatrix} + \frac{4E_c A_c}{mLH} \cos^3 \alpha N N^T$$

$$Q = \frac{2P_w}{mL} N_v, \quad q = \begin{Bmatrix} q_1 \\ q_2 \\ \vdots \\ q_n \end{Bmatrix}, \quad N = \begin{Bmatrix} \sin \frac{\pi}{2} \\ \sin \frac{2\pi}{2} \\ \vdots \\ \sin \frac{n\pi}{2} \end{Bmatrix}, \quad N_v = \begin{Bmatrix} \sin \frac{\pi vt}{L} \\ \sin \frac{2\pi vt}{L} \\ \vdots \\ \sin \frac{n\pi vt}{L} \end{Bmatrix} \quad (32)$$

3.4



[4]

TMD King-Post 가
3.3

3.2

$$M\ddot{q} + C\dot{q} + Kq = Q \quad (33)$$

$$M = \begin{bmatrix} I_{m \times n} & M_{12} \\ \mathbf{0}_{1 \times n} & m_z \end{bmatrix} \quad C = \begin{bmatrix} C_{11} & \mathbf{0}_{n \times 1} \\ C_{21} & c_z \end{bmatrix} \quad K = \begin{bmatrix} K_{11} & \mathbf{0}_{n \times 1} \\ K_{21} & k_z \end{bmatrix}$$

$$M_{12} = \frac{2m_z}{ml} N$$

$$C_{11} = \text{diag}(2\zeta_i \omega_i) \quad C_{21} = -c_z N^T \quad (i = 1, 2, \dots, n)$$

$$K = \text{diag}(\omega_i^2) + \frac{4E_c A_c}{mLH} \cos^3 \alpha NN^T \quad K_{21} = -k_z N^T \quad (i = 1, 2, \dots, n)$$

$$Q = \frac{2}{ml} (P_w N_v + m_z g N)$$

$$q = \begin{Bmatrix} q_1 \\ q_2 \\ \vdots \\ q_n \\ z \end{Bmatrix} \quad N = \begin{Bmatrix} \sin \frac{\pi}{2} \\ \sin \frac{2\pi}{2} \\ \vdots \\ \sin \frac{n\pi}{2} \end{Bmatrix} \quad N_v = \begin{Bmatrix} \sin \frac{\pi vt}{L} \\ \sin \frac{2\pi vt}{L} \\ \vdots \\ \sin \frac{n\pi vt}{L} \end{Bmatrix} \quad (34)$$

4.

4.1

1

TMD+King-Post

가

TMD, King-Post,

가

4.2

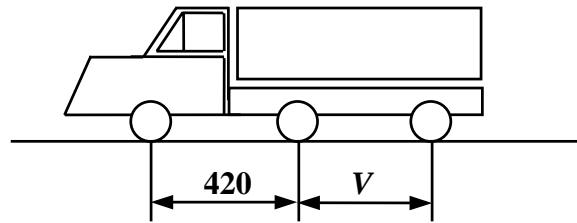
DB

[1]

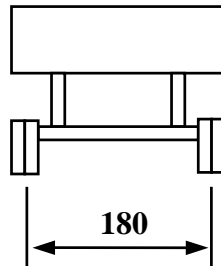
[5] DB

[1] DB

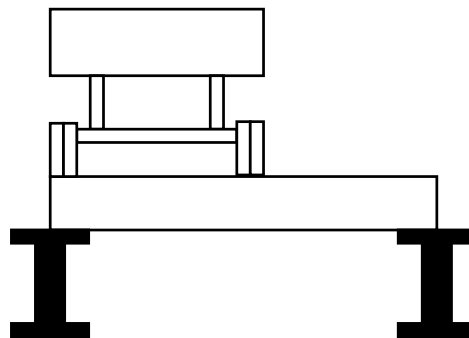
	W(tonf)	1.8W(tonf)	0.1W(kgf)	1.4W(kgf)
1	DB-24	43.2	2,400	9,600
2	DB-18	32.4	1,800	7,200
3	DB-13.5	24.3	1,350	5,400



▬ 0.1W ▬	▬ 0.4W ▬	▬ 0.4W ▬
0.1W	0.4W	0.4W



[5] DB



[6]

- [5] V 420~900 cm 가 580
 cm . 1 DB-24
 . 가
 420, 580 cm
 . 가
 [6] 1-0

4.3

4.3.1

TMD 3 ,
 가
 . TMD
 (multiple TMD)
 TMD 가 가 , Den
 Hartog 가 가 TMD
 가
 ,
 가 . Den Hartog 가

$$\varepsilon_z = \frac{m_z}{mL}, \quad \omega_z = \frac{\omega_n}{1 + \varepsilon_z} \quad (35)$$

$$\left(\frac{c_z}{c_c}\right)^2 = \frac{3\varepsilon_z}{8(1+\varepsilon_z)^3}, \quad c_c = 2m_z\omega_n \quad (36)$$

ε_z TMD, ω_z, m_z TMD, ω_n , c_z TMD, c_c TMD.

TMD

TMD

beating

. Tsai

(Critical Damper Damping)

가

(37)

beating

가

$$\zeta_z = \zeta_n + \sqrt{\varepsilon_z} \quad (37)$$

ζ_z, ζ_n TMD

가 (37)

가

TMD

beating

가

가

(37)

가

. TMD

,

가

가

4.3.2 King-Post

King-Post

Post

truss

. Post

Post

. King-Post

Post

truss

truss

[2]

[2] truss

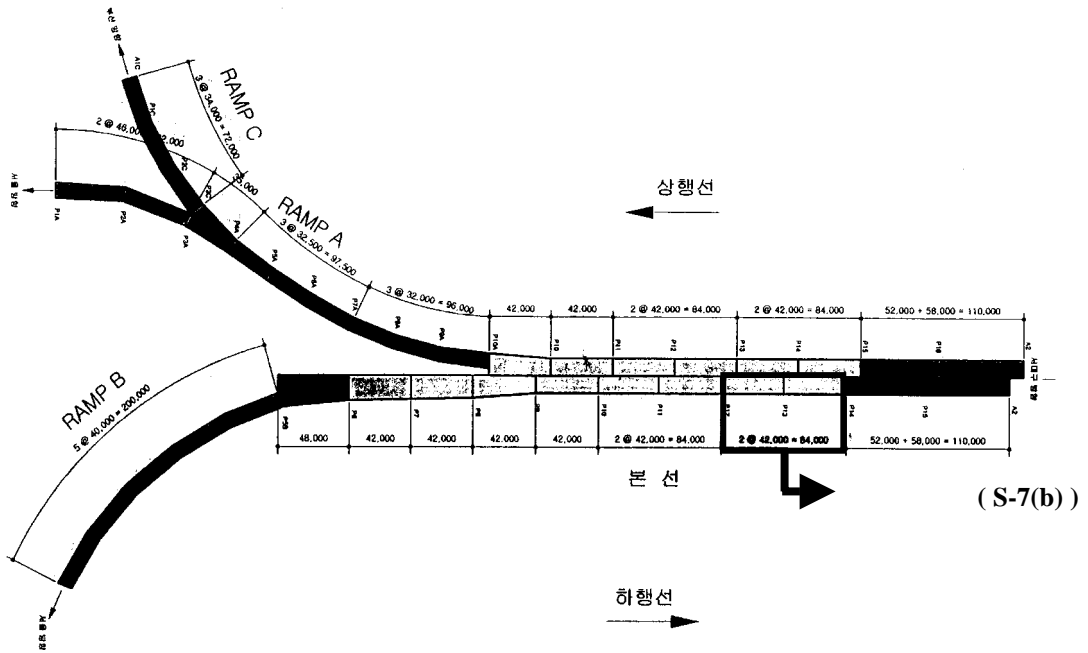
	$E_c (N / m^2)$	2.1×10^{11}
	$A_c (m^2)$	0.002
	$H (m)$	4

4.4

P.C. beam

694m

[7]



[7]

S-7(b)

42m

가

가

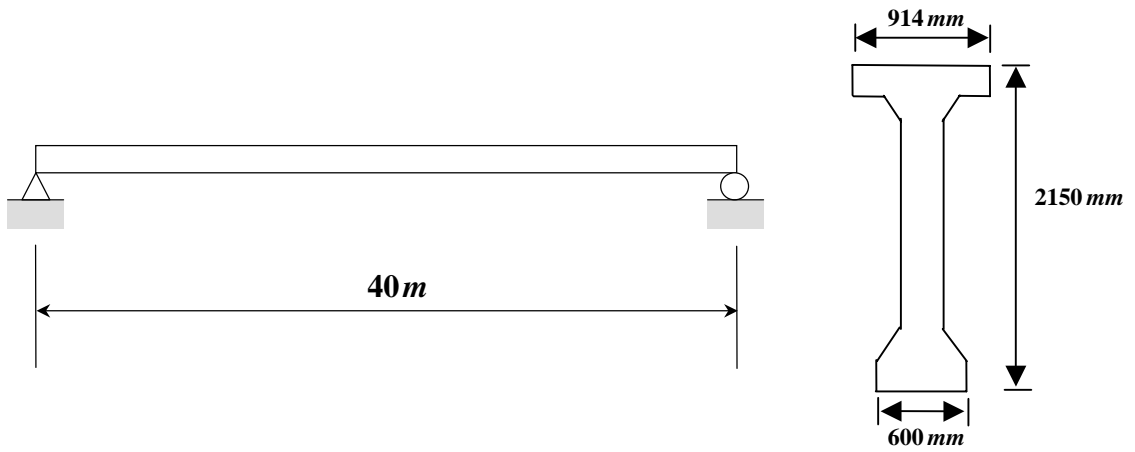
, PC

Bernoulli-Euler

[8]

[3]

, 가



[8]

[3]

	$2.94 \times 10^{10} \text{ N/m}^2$		0.3162 m^2
	2096.8 kg/m^3		0.1551 m^4
	0.06 %		

4.5

Newmark β

가 (Constant-Average-Acceleration Method)

$$\Delta t \quad (38)$$

$$\delta = \frac{1}{2}, \quad \alpha = \frac{1}{4}, \quad \Delta t = 0.01 \text{ sec} \quad (38)$$

[9]

5

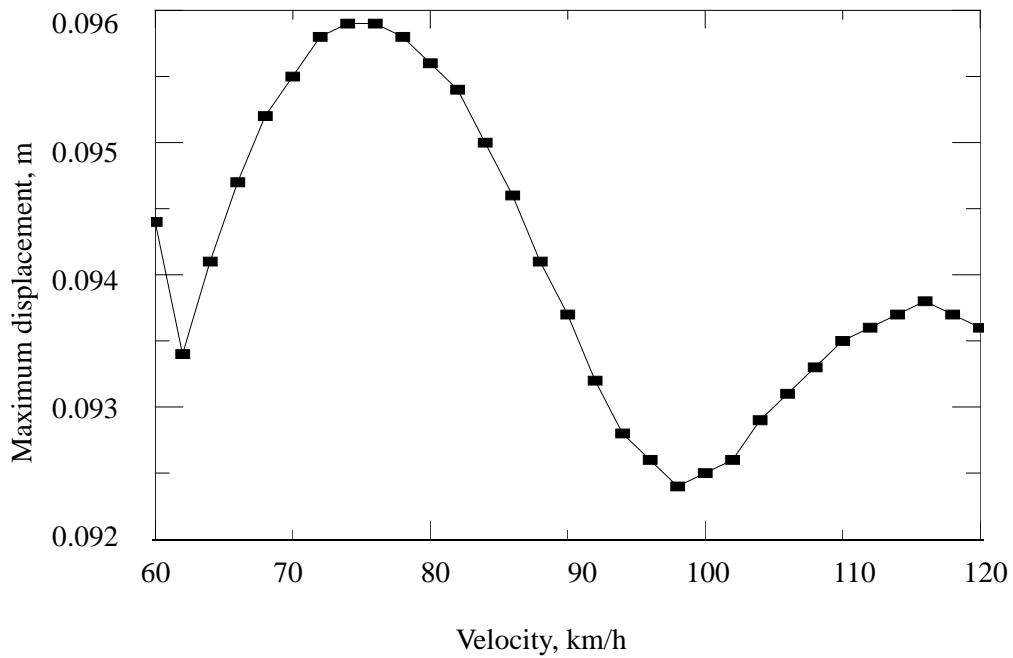
[9]

74 km

가 가

가

74km



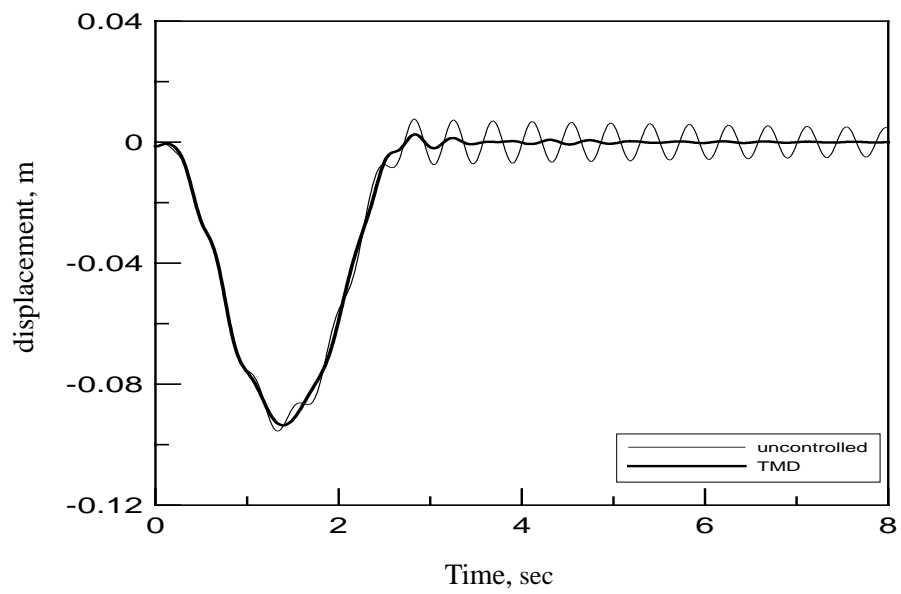
[9]

4.6

가 TMD, King-Post,

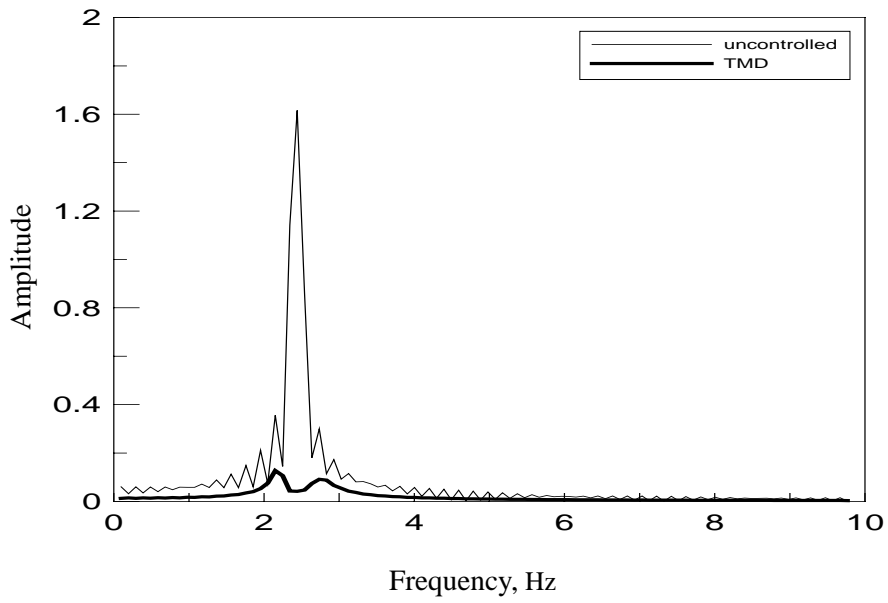
4.6.1 TMD

1)



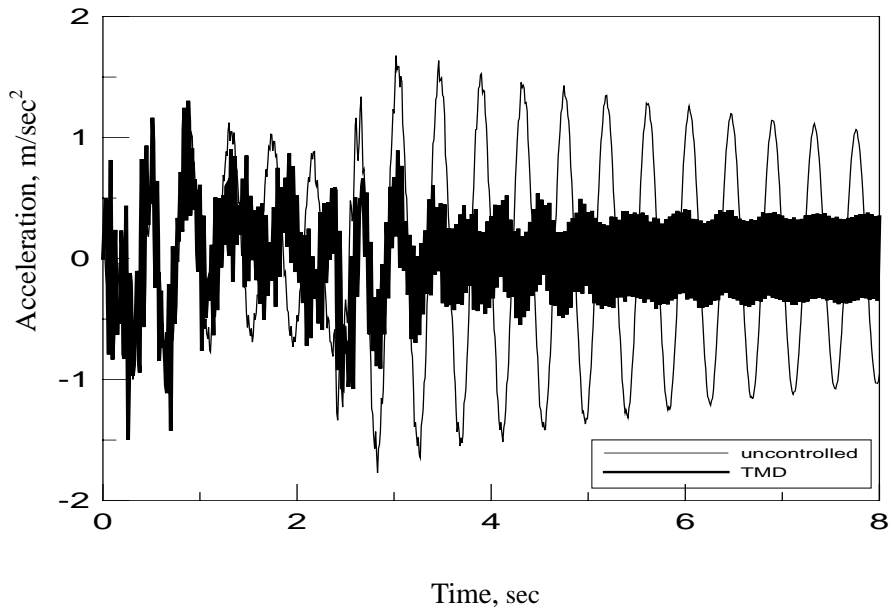
[10] TMD

[10] 가
0.0955 m, TMD 0.0936 m
2 % 가
[10] TMD
TMD
FFT(Fast
Fourier Transform)



[11] TMD FFT

2) 가



[12] TMD 가

가	[12] TMD 가	Root mean square(RMS) 가	가	TMD 가
1.4496 m/sec ²	0.8443 m/sec ²	1.7710m/sec ²	15.3 %	55.22 %

3) TMD

TMD

TMD
가

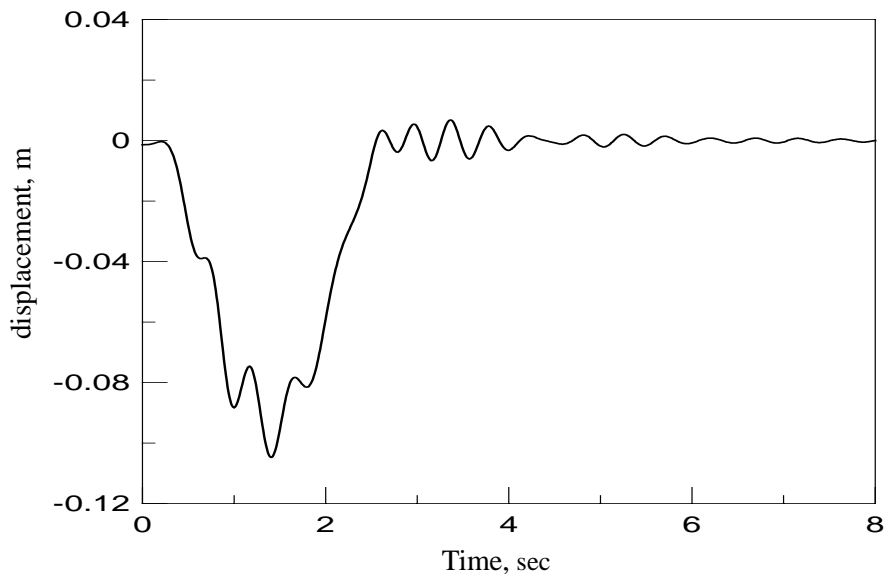
TMD

가

[13]

TMD

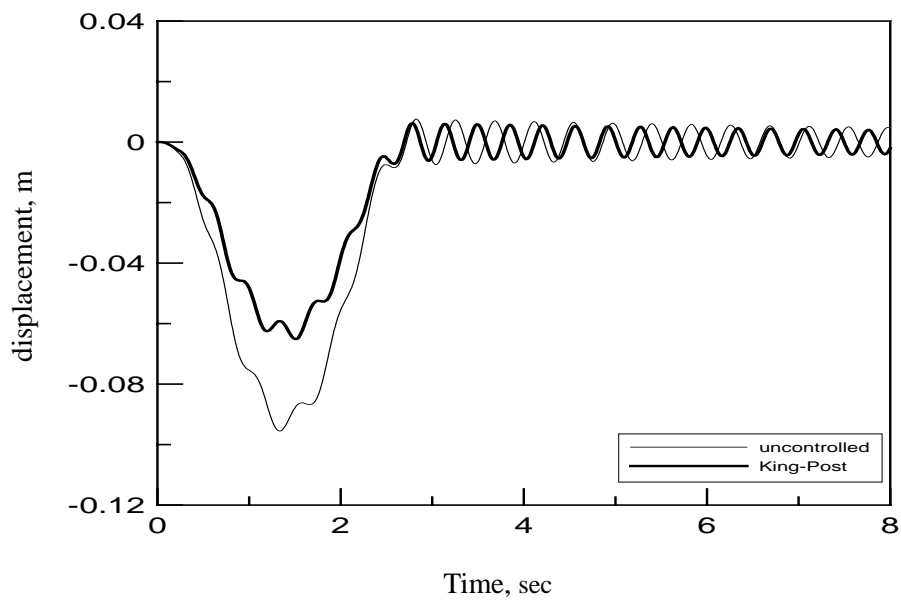
11.8%



[13] TMD

4.6.2 King-Post

1)



[14] King-Post

[14] King-Post

. King-Post

TMD

가

king-Post

0.0651 m

31.8 %

. King-Post

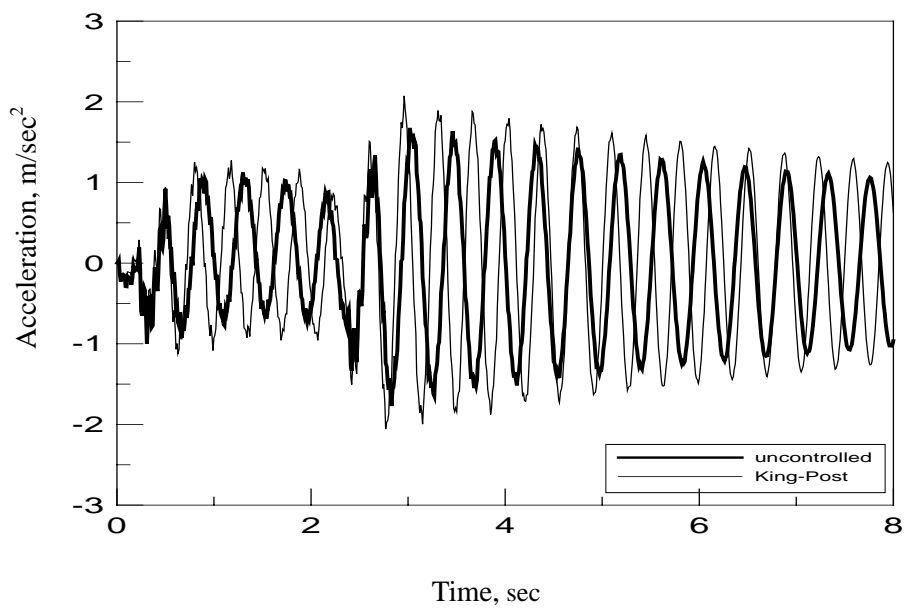
가

가

. King-Post

가

2) 가



[15] King-Post

가

[15] King-Post

가

King-Post

가

가

가

1.7710 m/sec²

가

2.0733

m/sec²

17 %

가가

. RMS

1.0049 m/sec²

19.02 %

가가

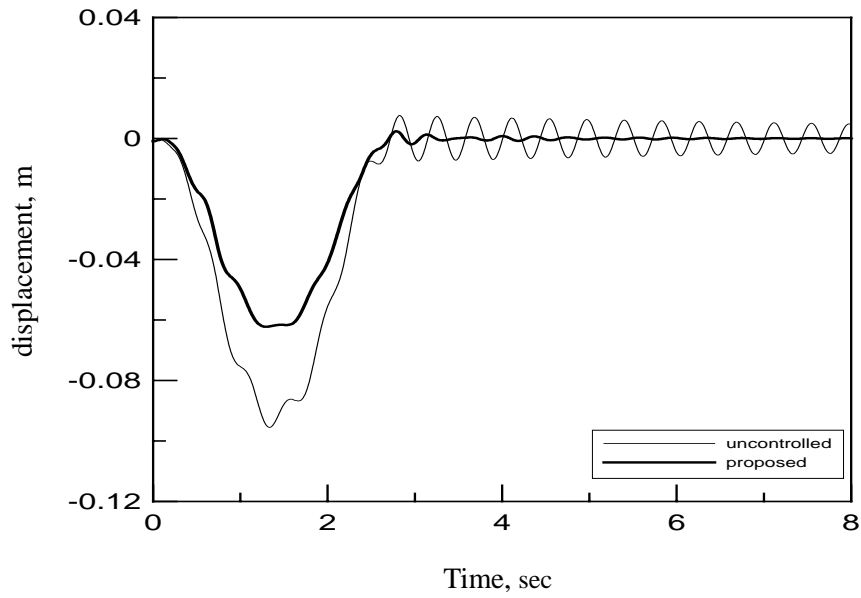
. King-Post

, King-Post

가

4.6.3

1)



[16]

가 [16]

0.0622m

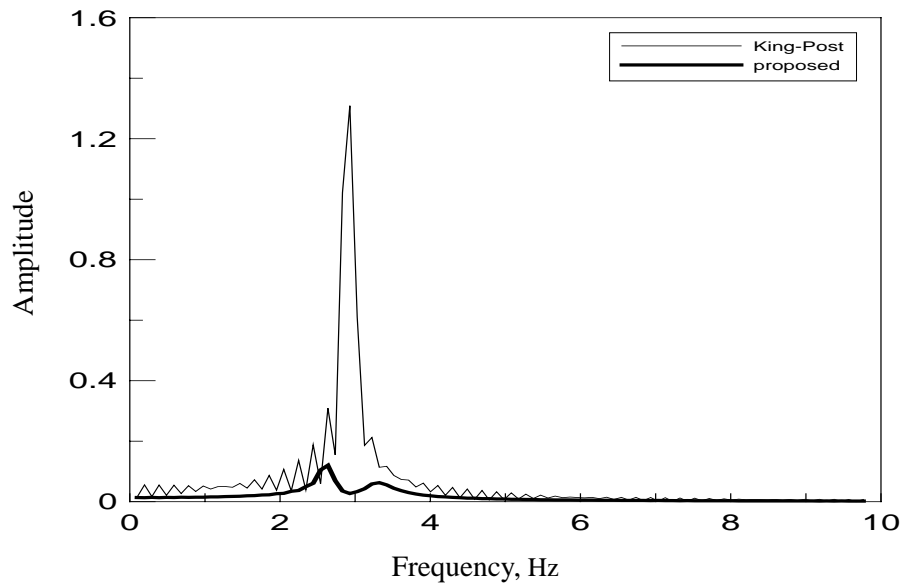
34.9%

가

TMD

King-Post

FFT(Fast Fourier Transform)



[17]

FFT

2) 가

[18]

. TMD

가

가

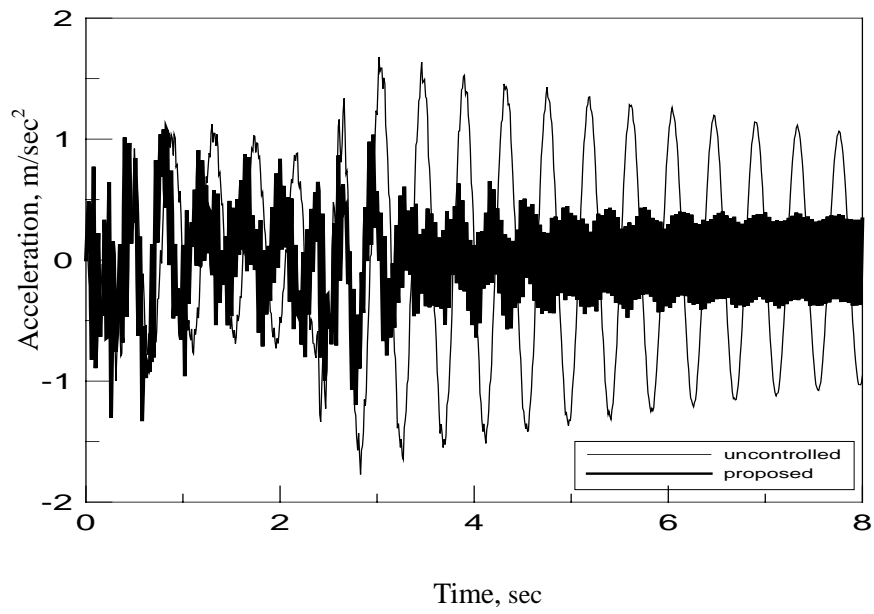
가

가

가

1.7710 m/sec²

가	1.3310 m/sec ²	24.8 %	가	가
RMS	0.8448 m/sec ²	0.3730 m/sec ²	55.82%	



[18]

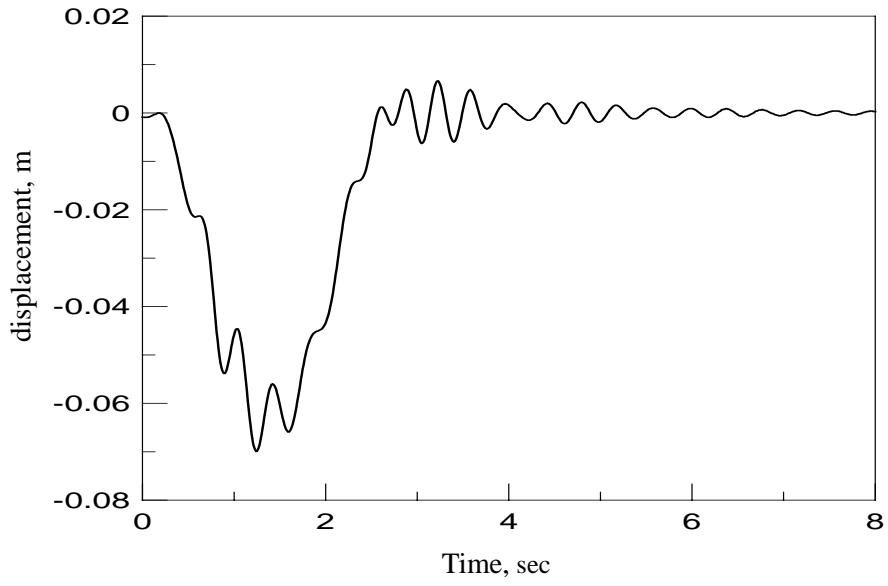
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3) TMD

[19]

TMD

12.4%



[19] TMD

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[4]

	uncontrolled	TMD	King-Post	Proposed
(m)	0.0955	0.0936 (-2%)	0.0651 (-31.8%)	0.0622 (-34.9%)
가 (m/sec ²)	1.7710	1.4996 (-15.3%)	2.0733 (+17%)	1.3310 (-24.8%)
RMS (m/sec ²)	0.8443	0.3781 (-55.22%)	1.0049 (+19.02%)	0.3730 (-55.82%)

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5.1

King-Post

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king-post

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5.2

Moving Force Model 가

Moving Mass Model

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31

1996.3 – 2001.2
2001.2 – 2003.2

(B.S.)

(M.S.)